

# AD A104325



## A GROUND-MOTION DISPLACEMENT GAUGE

Jack E. Whitener

R & D Associates

P. O. Box 9695

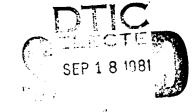
Marina del Rey, California 90291

1 August 1980

Topical Report for Period 5 January 1980-1 August 1980

CONTRACT No. DNA 001-80-C-0069

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED.



IG FILE COPY

THIS WORK SPONSORED BY THE DEFENSE NUCLEAR AGENCY UNDER RDT&E RMSS CODE B310080464 P99QAXDB00157 H2590D.

Prepared for

Director

DEFENSE NUCLEAR AGENCY

Washington, D. C. 20305

Destroy this report when it is no longer needed. Do not return to sender.

PLEASE NOTIFY THE DEFENSE NUCLEAR AGENCY, ATTN: STTI, WASHINGTON, D.C. 20305, IF YOUR ADDRESS IS INCORRECT, IF YOU WISH TO BE DELETED FROM THE DISTRIBUTION LIST, OR IF THE ADDRESSEE IS NO LONGER EMPLOYED BY YOUR ORGANIZATION.

## UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE					
(11) REPORT DOCUMENTATION	READ INSTRUCTIONS BEFORE COMPLETING FORM				
DNA 5574T	AD-A104 3	3. RECIPIENT'S CATALOG NUMBER 25			
A GROUND-MOTION DISPLACE	( //	Topical Report for  Period Stanto-laugeo /  6 PERIOR MINISTRACTOR APPENDIX NUMBER (5)  RDA-TR-113204-004			
7. AUTHOR(s)		8: CONTRACT OR GHANT NUMBER(s)			
Jack E. Whitener		DNA_001-80-C-0069			
9. PERFORMING ORGANIZATION NAME AND ADDRESS R & D Associates P. O. Box 9695 Marina del Rey, Californ: 11. CONTROLLING OFFICE NAME AND ADDRESS		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS Subtask P99QAXDB001-57			
Director Defense Nuclear Agency	1/	1 August 1980			
Washington, D.C. 20305		52 / 1/2/			
14. MONITORING AGENCY NAME & ADDRESS (If different	ent from Controlling Office)	UNCLASSIFIED			
		159. DECLASSIFICATION/DOWNGRADING SCHEDULE			
16. DISTRIBUTION STATEMENT (of this report)  Approved for public rele					
17. DISTRIBUTION STATEMENT (of the abstract entered of	in block 20, it different from rep	ort)			
This work sponsored by t RDT&E RMSS Code B3100804					
Displacement gauge Ground motion Particle displacement	Time-of-free Displacement Air drag				
of a mass released inside	ound-motion disp oncept is based a gauge canniste e-of-fall of the acement, togethe , uniquely deter mulation of grou	upon the free-fall time r prior to the onset mass to the cannister r with the free-fall mines the vertical dis-nd displacement by gauge			

DD FORM 1473

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE

EDITION OF 1 NOV 65 IS OBSOLETE

	UNCLASSIFIED	
SECURITY CLASS	FICATION OF THIS PAGE(When Data Entered)	
ł		
<b>[</b>		
1		
1		
Ì		
ţ		
<b>S</b>		
ł		
1		
}		
1		
(		
1		
Ì		
l		
}		
}		
ł		
1		
Ì		
[		
İ		
}		
1		
ì		
· ·		
}		
l		
<b>\</b>		
§ .		
1		
{		
j		
1		
1		
1		
<b>(</b>		
}		
Ì		
L		

UNCLASSIFIED

#### SUMMARY

This report summarizes the concept, applications, and results of developmental tests for a special ground-motion displacement gauge. Other possible uses, which have not been tested, are also discussed.

The gauge concept is based upon the free-fall time of a mass released inside a gauge cannister prior to the onset of ground motion. The time-of-fall of the mass to the cannister bottom during ground displacement, together with the free-fall time for the gauge at rest, uniquely determines the vertical displacement at one time. Simulation ( ground displacement by gauge drop tests demonstrated the operating principle.

One application of the gauge, for which it was originally proposed, would be to measure permanent displacements produced by underground explosions. The displacement potential determined from the measurement would provide a source strength of seismic signals from the explosion. Another use is to verify the measurements of ground displacement obtained by integration of particle velocity or acceleration gauge data. The size of the gauge is one limiting factor on the range of displacement which can be measured. Another is the ground stress which the gauge must survive. The practical range of measurements would be from .1- to 100-cm displacement.

## PREFACE

This report describes a ground-motion displacement gauge previously proposed by RDA. Publication of the report was prompted by suggestions from Mr. R. J. Port of RDA for new applications in high-explosive simulation programs. Dr. E. A. Martinelli offered valuable assistance on other applications discussed in the report.

## TABLE OF CONTENTS

Section		Page
	SUMMARY	1
	PREFACE	2
	LIST OF ILLUSTRATIONS	5
I	INTRODUCTION	7
II	DESCRIPTION OF THE VERTICAL DISPLACEMENT GAUGE	8
	1. Operating Principle	8
	2. Operational Constraints	11
	<ol> <li>Review of Advantages and Disadvantages of the Gauge Design</li> </ol>	18
	4. Sources of Error in Measurements	19
III	DEVELOPMENT AND TESTS OF PROTOTYPE DISPLACMENT GAUGE	E- 20
	1. Gauge Construction	20
	<ol><li>Laboratory Tests and Discussion of Results</li></ol>	23
IV	MODIFICATION OF GAUGE DESIGN FOR OTHER USES	28
	l. Particle Velocity	28
	2. Horizontal and Vertical Displacement	31
V	VERIFICATION TESTS OF ACCELEROMETER AND VELOCITY GAUGES	34
	REFERENCES	36

## TABLE OF CONTENTS (CONCLUDED)

Section			Page
	APPENDIX A.	DISPLACEMENT POTENTIAL	37
	APPENDIX B.	EFFECTS OF AIR DRAG ON DIS- PLACEMENT MEASUREMENTS	39
	APPENDIX C.	ERROR DUE TO ROTATIONAL	43

## LIST OF ILLUSTRATIONS

Figure		Page
1	Principle of ground-motion gauge operation	9
2	Schematic showing basic components of ground-motion displacement gauge	10
3	Estimate of gauge size vs zero-displace- ment drop time	12
4	Illustration of displacement at impact time for varying gauge sizes and waveforms	13
5	Effects of gauge size and weight-release time on recorded upward displacement	14
6	Effects of mass-release time and gauge size upon measured displacement of a down-ward vertical motion	16
7	Particle displacement at 12-m range and 9-m depth below ground surface from a 500-ton surface burst	17
8	Full-scale drawing of prototype ground- motion displacement gauge	21
9	Block diagram of displacement gauge system	22
10	Schematic diagram of displacement gauge circuitry	24
11	Experimental setup for simulation tests of displacement gauge	27
12	Modified displacement velocity gauge	29
13	Particle velocity obtained from displace- ment measurements	30
14	Schematic illustration of technique to measure horizontal and vertical displace-ment	32
15	Ground motions vs time, Mineral Rock Gauge 60-18	35
Al	Displacement potential obtained from permanent-displacement measurement	40
Cl	Illustration of effects on ground-motion displacement measurements due to rotational motion	
C2		46
C2	Displacement error due to gauge rotation	48

## LIST OF TABLES

<u> Table</u>		Page
1	Free-fall drop-test measurements	25
2	Drop-test measurements	26
в1	Percentage errors in displacement due to air drag	44

#### I. INTRODUCTION

This report describes the concept and development of a ground-motion displacement gauge suggested by the author at RDA in 1972 (Ref. 1). During the period 1972 through 1973, the MICHTY MITE series of nuclear and high-explosive underground tests were conducted at the Nevada Test Site. The objective of these tests was to determine the effectiveness of various heat sink concepts for quenching explosions inside cavities. The Advanced Research Projects Agency (ARPA) sponsored the test program which was executed by the Defense Nuclear Agency. ARPA was primarily concerned that, in the event of complete test ban negotiations, clandestine tests in underground cavities could deny seismic detection. Reference 2 describes these tests.

From the theory of cavity decoupling, it has been shown that the seismic energy transmitted to teleseismic distances ( $\sim 10^3$  km) is related to the permanent elastic particle displacement at or near the cavity wall (Ref. 3).

In Appendix A a derivation is given for the relation between permanent displacement and the energy released in a cavity. This application of a permanent ground-motion displacement measurement motivated the concept of the gauge discussed in this report.

#### II. DESCRIPTION OF THE VERTICAL DISPLACEMENT GAUGE

#### 1. OPERATING PRINCIPLE

The principle of gauge operation is illustrated using Figures 1 and 2. In Figure 1, the initial position of the gauge case is shown in the solid figure. Any point A of the gauge case moves to a new position B along the displacement path. At time prior to the beginning of motion, a mass is released at the top of the gauge. On impact of the mass shorting pin upon the foil switch located at the bottom of the gauge case, the time of impact is recorded with the gauge now at position B. During the time-of-fall, the mass drops through the distance & and the gauge is displaced vertically by the amount d as shown. For the case of no vertical motion, the mass would drop a fixed distance S as shown in Figure 1. From the recorded release and impact times, t and t respectively, the displacement is determined by

$$d = S - 1/2g(t_i - t_r)^2$$

or alternatively,

$$d = 1/2g \left\{ t_0^2 - (t_i - t_r)^2 \right\}$$
 (1)

where

$$S = 1/2gt_0^2 \tag{2}$$

and  $t_0$  is the drop time for zero displacement. If  $t_i$  -  $t_r$  >  $t_0$ , the displacement is downward. If  $t_i$  -  $t_r$  <  $t_0$ , the motion is upward.

It is emphasized that so far this gauge design measures only vertical displacement at just one point.

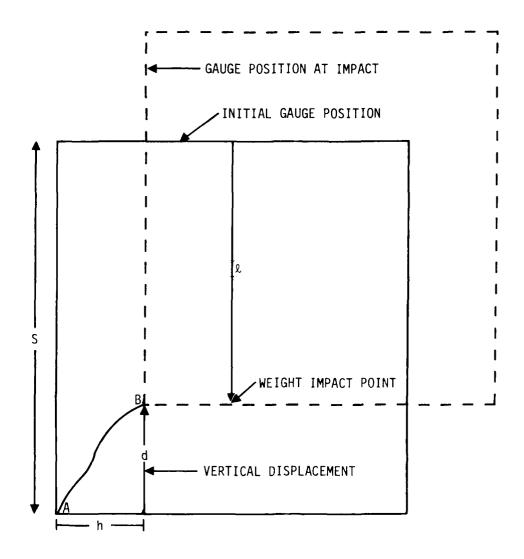


Figure 1. Principle of ground-motion gauge operation.

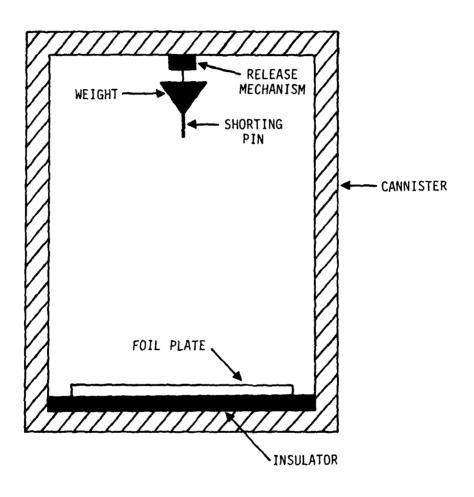


Figure 2. Schematic showing basic components of ground-motion displacement gauge.

## 2. OPERATIONAL CONSTRAINTS

Several features of this technique may limit the application of the gauge to a specific range of displacement measurements. Gauge construction must be preceded by careful estimates of the maximum motions and durations expected.

## a. Gauge dimensions

(1) Upward displacement--The maximum upward displacement which can be measured must be less than S, otherwise the weight impact would occur prior to the peak value of displacement. Thus the vertical dimensions must be greater than 1/2gt<sub>o</sub><sup>2</sup>. Figure 3 shows a plot of Eq. 2 to estimate the gauge size S. It is noted that the gauge size rapidly increases with the time-of-fall for upward motions.

In Figure 4, two upward-displacement waveforms are shown corresponding to a cube-root yield scaling factor of 2. The dashed curves are the free-fall paths for the different gauge sizes shown and for zero time of release. A mass drop height of almost 8 cm is required to measure displacement beyond the peak for W = 1, while about 4 cm is required for the smaller waveform. For smaller drop heights S, impact occurs prior to the peak displacement. It can also be noted that the gauge size S does not scale with  $W^{1/3}$  since the time-of-fall of the weight does not scale.

Another parameter which could be useful is controlled advance or delay time-of-weight release. Figure 5 illustrates these effects of advance or delay in the release time relative to zero time of release for upward displacement. The figure shows the different impact times for releases at -15 and +15 ms for 2.54-cm and 5-cm gauge sizes.

(2) <u>Downward displacement</u>--Downward displacements present different problems since the gauge motion is following the free-fall of the weight. The downward displacement

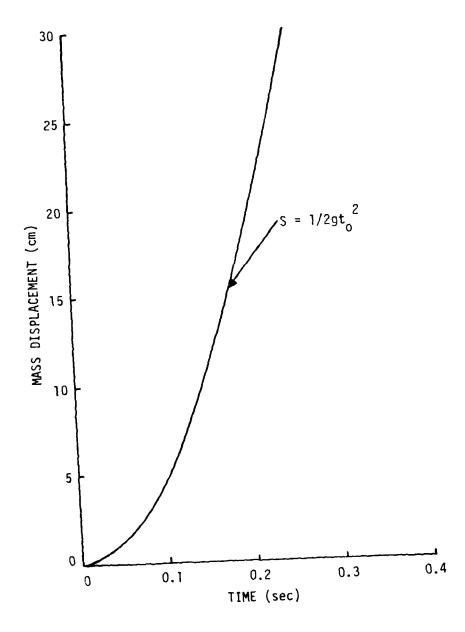


Figure 3. Estimate of gauge size vs zero-displacement drop time.

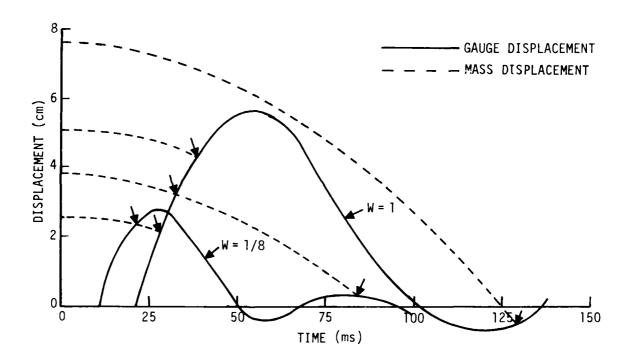


Figure 4. Illustration of displacement at impact time for varying gauge sizes and waveforms.

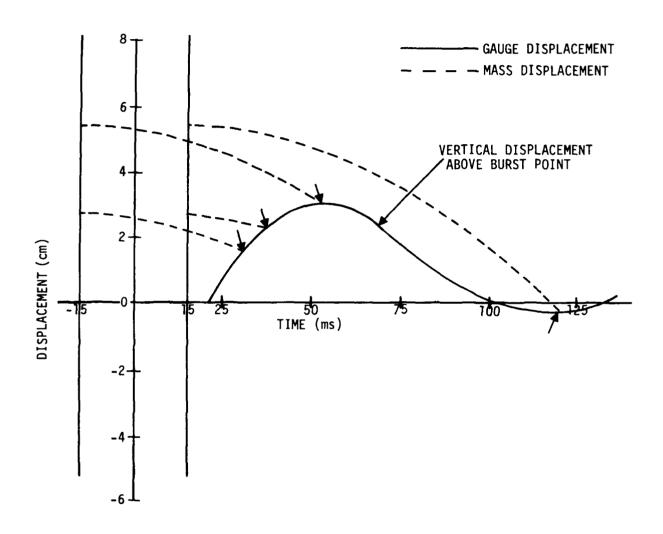


Figure 5. Effects of gauge size and weight-release time on recorded upward displacement.

d(t) at any time t must not exceed the free-fall position of the mass; namely,

$$d(t) \le 1/2g(t - t_r)^2$$

Otherwise, the top of the gauge case would overtake the falling weight.

Several different modes of gauge operation will affect the measured displacement for downward motion. Figure 6 illustrates these cases. Traces of the particle motion of the top and bottom points of the gauge are shown.

- Curve A illustrates a case where the gauge overtakes the mass released at zero time, which would result in failure of the measurement.
- Curve B shows the effect of pre-release of the mass at -25 ms and for the same initial mass height of 2.54 cm. In this case, the weight trajectory does not intersect that of the gauge top.
- Curve C shows the effect of reducing the gauge size from 5 cm to 2.5 cm and pre-release of the weight at -50 ms.
- Figure 1, the horizontal gauge displacement h is shown at position B. From the figure it can be seen that the maximum horizontal displacement at impact must be less than the initial distance to the gauge wall. Otherwise, the mass would strike the wall and interfere with the time-of-fall measurement. The horizontal dimensions then must be taken into account in the specific application of the gauge. Figure 7 illustrates the complicated particle motion below the ground surface from a surface burst. The gauge would require more than 5 cm

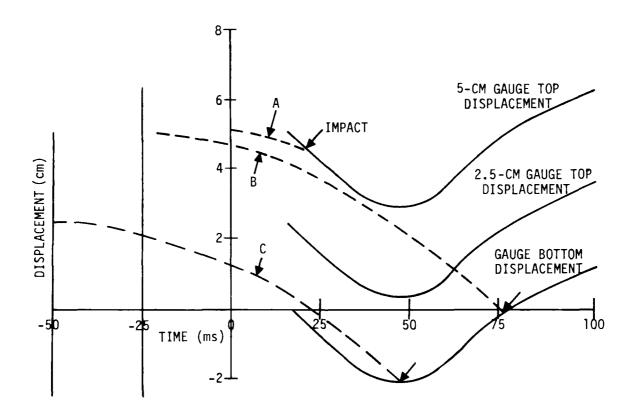
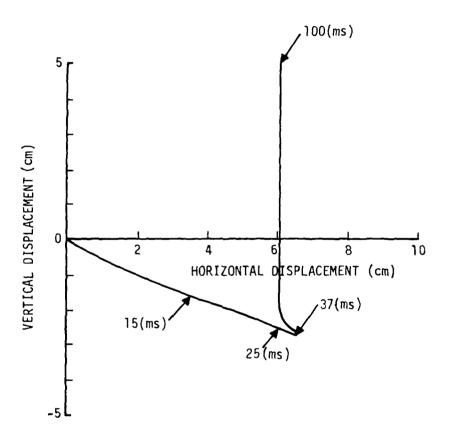


Figure 6. Effects of mass-release time and gauge size upon measured displacement of a downward vertical motion.



The state of the s

Figure 7. Particle displacement at 12-m range and 9-m depth below ground surface from a 500-ton surface burst.

horizontal clearance between the initial weight position and the gauge wall to record vertical motion to times greater than 37 ms.

- (4) Gauge case strength--Mechanical hardening requirements for the displacement gauge may be greater than those for other types of ground-motion gauges. The large void required inside the gauge casing to accommodate the horizontal and vertical motions of the falling weight will increase the structural strength necessary to prevent buckling or crushing of the case prior to the measurement. The stress level at which a desired displacement measurement is to be made will depend upon the source yield and the peak displacement. All these factors must be taken into account in the design of the gauge housing. A great deal of experimental data on shock loading of cylindrical and other structures is available to estimate the structural hardening requirements or limitations for the design of the displacement gauge.
- 3. REVIEW OF ADVANTAGES AND DISADVANTAGES OF THE GAUGE DESIGN

These are some of the advantages associated with the gauge design described:

- No mechanical or electronic integration is required.
- There is an accurate inertial frame of reference (free fall).
- The data recorded includes only time-interval measurements, while amplitude measurements require calibration.
- Time-base drift is not possible.
- Time measurements of switch opening and closure provide high signal-to-noise ratio.

This gauge design does have certain disadvantages, such as:

- Only the vertical component of displacement is possible in the present gauge design.
- The dependence of gauge size upon magnitude of displacement requires careful estimates of expected ground motions.
- The time-squared dependence of free-fall distance imposes practical limitations on gauge size.

#### 4. SOURCES OF ERROR IN MEASUREMENTS

Two possible effects on the accuracy of measurement are considered.

- Air drag on drop mass. Relative motion of the drop mass and the air enclosed in the gauge cannister will induce air drag, which will oppose the free-fall motion of the weight. In Appendix B, this problem is analyzed with estimates of errors expected.
- Rotational motion effects on measurements. If the gauge motion undergoes a rotation, the effect on the measurement is to introduce an error in the measurement. In Appendix C, this effect is discussed.

#### III. DEVELOPMENT AND TESTS OF PROTOTYPE DISPLACEMENT GAUGE

#### 1. GAUGE CONSTRUCTION

A prototype displacement gauge was built and tested by Systems, Science and Software at the Green Farm Test Facility in March 1974. Specific design features of the gauge were contributed by the author and Dr. Howard Kratz (formerly of Systems, Science and Software). Electronic circuitry design and laboratory testing were performed by Mr. William Muhl of IMED Corporation (Ref. 4). Figure 8 is an engineering sketch of the full-scale model with specific features of the gauge indicated.

- a. Mass-release mechanism—The drop mass is supported by the copper wire under tension from the leaf spring. An electrical circuit (normally closed) is completed through the breakwire to the drop mass and mass support. On receiving a release signal, the solenoid is energized, breaking the wire which opens the electrical circuit and at the same time releases the drop mass. Opening of the circuit supplies an electrical timing signal of the mass release.
- b. Foil switch-The foil switch, located at the bottom of the gauge case, consists of a copper foil insulated from the metal case by a mylar foil. An electrical circuit (initially open at the switch) is completed by the shorting pin attached to the drop mass, generating an electrical timing signal at impact.
- c. <u>Electrical circuitry</u>—A block diagram of the electrical circuitry is shown in Figure 9. The mass-release delay circuit generates a signal prior to shock arrival at the gauge. Timing of the mass release could occur at or prior to detonation time, or be delayed depending upon the distance between the gauge and explosive charge.

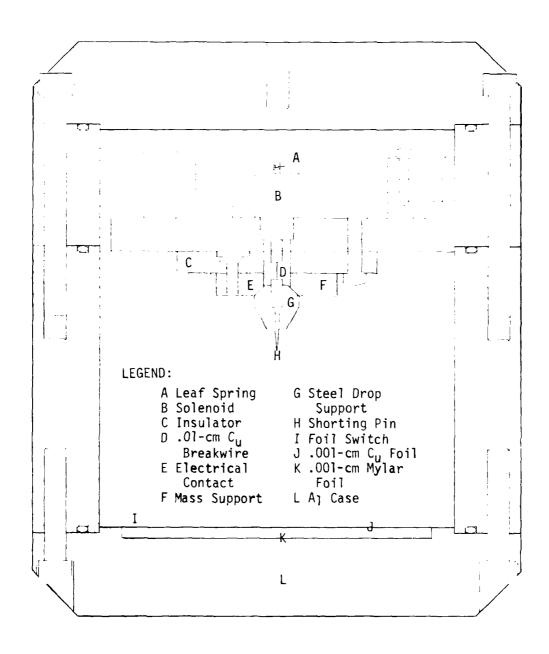


Figure 8. Full-scale drawing of prototype ground-motion displacement gauge.

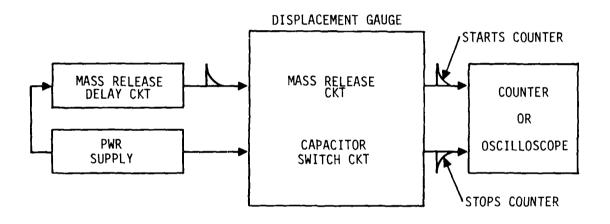


Figure 9. Block diagram of displacement gauge system.

The signal generated by the breakwire initiates the horizontal sweep of an oscilloscope, or can be used to start a counter. Arrival of the impact signal is recorded either on the oscilloscope or stops the counter if used. The time difference provides the time-of-fall from which the vertical displacement is calculated. Figure 10 is a schematic diagram showing the signal-generating circuits.

## 2. LABORATORY TESTS AND DISCUSSION OF RESULTS

During the MIGHTY MITE series of underground tests on seismic decoupling (Ref. 1), a test was planned that included the displacement gauge concept in the test program. That test was later cancelled; however, funding for construction and laboratory testing of the gauge was approved. Although no ground-motion displacement tests were attempted, gauge-displacement experiments were conducted to verify the gauge concept. Two sets of experiments are described and the results are presented in the following paragraphs.

a. <u>Time-of-fall measurements--The first series of experiments simply consisted of measuring the free-fall time of the drop weight through an accurately measured drop distance with no motion of the gauge case. All the electronic circuits were energized as previously discussed. A trigger signal applied to the weight-release mechanism started an electronic counter with a time resolution of a few µsec. A signal at impact stopped the counter from which the time-of-fall was recorded.

Table 1 summarizes these results obtained from the given drop distance. Also listed are the time differences from the calculated value, and the percentage differences. The consistency in the measurements would seem to indicate a systematic error either in the timing measurements or in the recorded drop distance. The difference between to and the mean is 2.0 ms.</u>

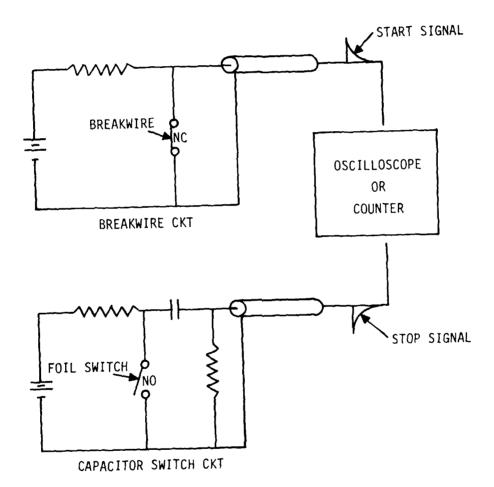


Figure 10. Schematic diagram of displacement gauge circuitry.

## TABLE 1. FREE-FALL DROP-TEST MEASUREMENTS

Drop Distance S = 3.498 cm Acceleration  $g = 980 \text{ cm/sec}^2$ Calculated Time-of-Fall  $t_0 = 84.486 \text{ ms}$ 

TEST NO.	TIME-OF-FALL (ms)	(t <sub>o</sub> - t)ms	% DIFF
1	82.52	1.967	2.32
2	82.64	1.845	2.18
3	82.21	2.276	2.69
4	82.70	1.786	2.11
5	82.69	1.796	2.13
6	82.60	1.886	2.23
7	82.22	2.266	2.68
8	82.09	2.396	2.84

 $<sup>\</sup>overline{t}$  = 82.46(ms)  $\sigma$  = 0.230(ms)

b. Gauge drop tests--To simulate ground-motion displacement tests, the gauge was suspended a measured distance  $\mathbf{d}_{\mathrm{m}}$  above a bench top as shown in Figure 11. The gauge cannister was released by firing of the HE detonator just after the drop mass was released. Measurements of time-of-fall  $\mathbf{t}_{\mathrm{m}}$  of the mass were recorded from which the apparent displacement was calculated by

$$d_a = S - 1/2gt_m^2$$

where S is the initial pin distance from the bottom of the gauge.

Table 2 summarizes the test results. The apparent displacements are significantly different from the displacements (gauge drop distances) measured prior to each test by more than 10 percent. Again, as noted in Table 1, the time-of-fall in each case is less than was predicted from the known displacement. Applying the correction of 2 ms from Table 1, the corrected percentages of difference in the displacements are reduced as shown in the last column of Table 2. These results do not determine the source of the apparent systematic error, and time did not permit further investigation of the problem. There is little reason to doubt that the accuracy of the measurements can be significantly improved by trouble-shooting the weight release and impact timing.

TABLE 2. DROP-TEST MEASUREMENTS

 $S = 5.08 \text{ cm}; \quad q = 980 \text{ cm/sec}^2$ 

TEST NO.	t <sub>m</sub> (ms)	d <sub>m</sub> (cm)	d <sub>a</sub> (cm)	% DIFF	% DIFF (corr)
1	123.575	2.774	2.403	13.4	4.58
2	123.294	2.703	2.368	12.5	3.36
3	122.060	2.530	2.220	12.2	2.70
4	125.090	2.891	2.5873	10.5	1.95

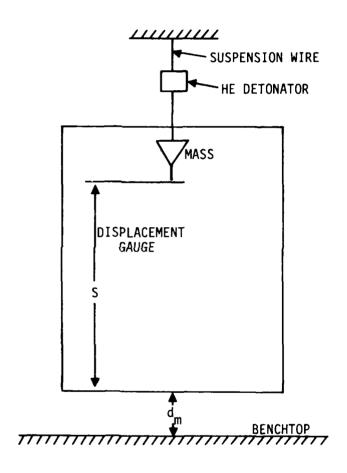


Figure 11. Experimental setup for simulation tests of displacement gauge.

#### IV. MODIFICATION OF GAUGE DESIGN FOR OTHER USES

#### PARTICLE VELOCITY

Measurement of particle velocity, in addition to particle displacement, would provide another data check on existing particle velocity gauges. The same principle of time-interval measurements is all that would be required using the concept that will be described in this section.

Figure 12 illustrates the concept. Two separate displacement gauges are contained within the same cannister as shown. Two drop masses and two impact switches would provide two near-displacement measurements separated by a small time interval. The spacing of the two measurements could be accomplished in either of two ways. Shorting pins of unequal length released simultaneously would provide two displacements. From Figure 12 the masses released at  $t_r$  would impact at times  $t_1$  and  $t_2$ . The corresponding displacements are given by

$$d_1 = s_1 - 1/2g(t_1 - t_r)^2$$

$$d_2 = S_2 - 1/2g(t_2 - t_r)^2$$
,

from which the average velocity over the displacement interval is obtained:

$$\overline{u} = \frac{d_2 - d_1}{t_2 - t_1}.$$

Figure 13 illustrates the principle of velocity measurement. Two different displacement curves are shown to illustrate positive and negative velocity measurements.

The second method works basically in the same manner where the two masses are released at different times,

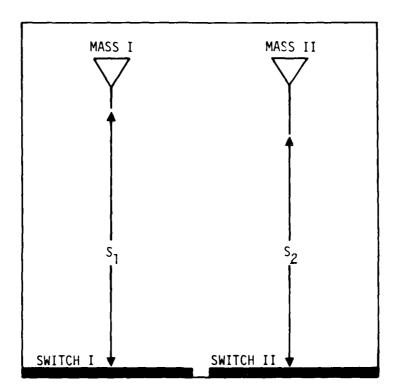


Figure 12. Modified displacement velocity gauge.

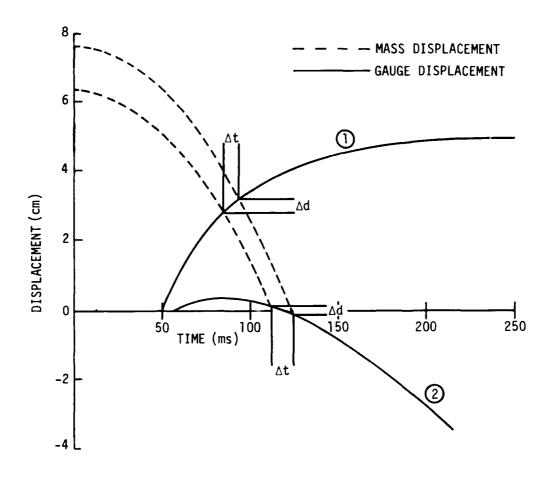


Figure 13. Particle velocity obtained from displacement measurements.

 $\mathbf{t_1}$  and  $\mathbf{t_2}$ , with impact times  $\mathbf{t_3}$  and  $\mathbf{t_4}$ , respectively. The displacements are obtained from

$$d_1 = S - 1/2g(t_3 - t_1)^2$$

$$d_2 = S - 1/2g(t_4 - t_2)^2$$

and the velocity calculated from

$$\overline{u} = \frac{d_2 - d_1}{t_4 - t_3}.$$

#### HORIZONTAL AND VERTICAL DISPLACEMENT

Another possible application of the gauge concept is the measurement of the horizontal and vertical components of permanent displacement. Only the vertical component can be measured using the basic design as illustrated in Figure 1. The horizontal motion could be inferred from the vertical measurement in the case of known spherical symmetry along a given radial direction.

A gauge of special design, used in conjunction with the basic model, could in principle be used to determine both components of permanent displacement. Figure 14 is an illustration of the dual gauge principle where Gauge I has been modified by constructing the foil switch in the form of a cone as shown. Gauge II is the basic design discussed throughout the report. Operation of the system is as follows: at 'zero time' both weights are released at a distance S from the common base BB. The gauge undergoes a final upward displacement D, with horizontal and vertical components H and V as shown. Impacts of the masses at P<sub>1</sub> and P<sub>2</sub> occur at times t<sub>1</sub> and t<sub>2</sub> respectively, with the base now located at B'B'. Motions of the drop masses at impact are given by

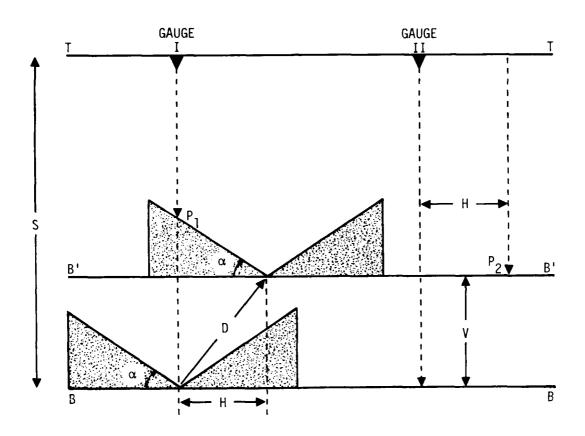


Figure 14. Schematic illustration of technique to measure horizontal and vertical displacement.

$$1/2gt_1^2 = S - V - Htan\alpha$$
  
 $1/2gt_2^2 = S - V$ ,

hence,

$$1/2g\left(t_2^2-t_1^2\right) = Htan\alpha,$$

where  $\alpha$  is the angle subtended from the horizontal by the conical foil switch. The displacement can then be calculated to be

$$H = 1/2g \left(t_2^2 - t_1^2\right) / tan\alpha$$
  
 $V = S - 1/2gt_2^2$ 

It should be noted that this technique can only be used for a final displacement since, in the case of a transient measurement, two different displacements would occur at times  $\mathbf{t}_1$  and  $\mathbf{t}_2$ , which would not uniquely determine these displacements.

## V. VERIFICATION TESTS OF ACCELEROMETER AND VELOCITY GAUGES

More recently, another application has been suggested. Particularly with the increase in the application of high explosives for the simulation of nuclear explosions, the gauge can be an important tool in calibrating ground-motion measurements.

Ground-motion displacement is generally obtained either by double time-integration of acceleration data or single time-integration of particle velocity data. This has been the case since no completely satisfactory displacement gauge has been developed. The measurement of particle displacement poses a difficult problem in devising schemes for direct measurement, since a fixed-reference position is required to determine the relative displacement. However, the methods using integration of acceleration and particle velocity data often introduce errors in the calculated displacement. reference voltage or signal output from a gauge may vary with time (baseline shift) which introduces a cumulative error of unknown value in the displacement obtained by integration. The total signal consists of the real signal superimposed upon the varying reference base. Figure 15 illustrates the different ground displacement results obtained from actual time-integrated accelerometer and particle velocity data. Both gauges were contained in the same instrumentation package. The solid curve shows the results from the acceleration data which have been singly integrated to obtain the particle velocity and doubly integrated to yield displacement. dotted curve is a record of the velocity measurement and its integration to obtain displacement. These results show the necessity for a method to measure displacement at a point to determine the discrepancy in the measurements.

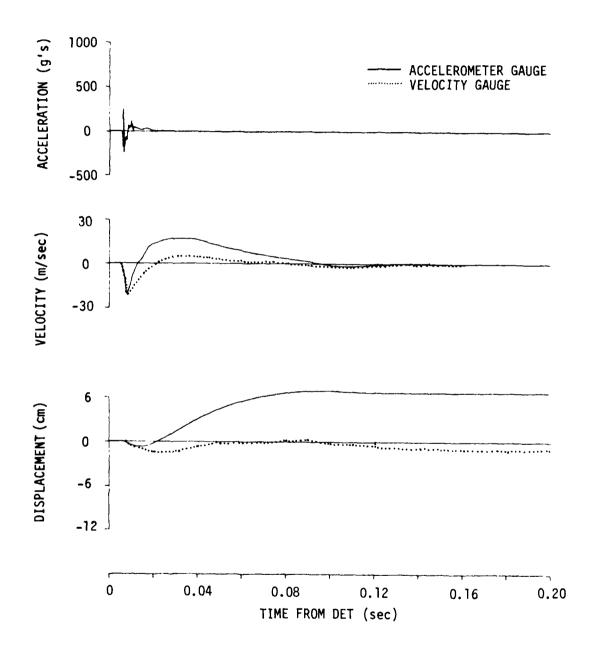


Figure 15. Ground motions vs time, Mineral Rock Gauge 60-18.

#### REFERENCES

- 1. Whitener, J. E., <u>Proposed Test Program to Study Mechanisms</u>
  Involved in Enhanced Cavity Decoupling, R & D Associates,
  RDA-TR-075-DNA, May 1972.
- 2. Whitener, J. E., MIGHTY MINE Series, (a) DIAMOND DUST

  Event , (Unpublished); (b) DIAMOND MINE Event
  (Unpublished); (c) PRE-MINE DUST Event, POR 6822,
  March 1975; Technical Directors Summary Reports, Defense
  Nuclear Agency.
- 3. Latter, A. L., et al., A Method of Concealing Underground Nuclear Explosions, The RAND Corporation, RM-2347, March 1959
- 4. Muhl, W., Laboratory Research Note Book, Systems, Science and Software, March 1974.
- 5. White, J. E., Seismic Waves: Radiation, Transmission, and Attenuation, McGraw-Hill, 1965.

# APPENDIX A. DISPLACEMENT POTENTIAL

From the theory of elasticity, the permanent elastic displacement of the cavity wall is related to the energy released by an explosion given by

$$\frac{4\pi}{3} \quad \frac{\text{Pa}^3}{\text{$\gamma$-1}} \quad = \quad \frac{16\pi\mu a^2 d_a}{3(\gamma-1)}$$

where p = cavity pressure,

 $\mu \equiv$  shear modulus of medium,

a = radius of the cavity,

 $d_a = permanent wall displacement.$ 

With the approximation of incompressible flow,

$$a^2d_a = r^2d_r ,$$

the effective energy released in the cavity becomes

$$\frac{4\pi}{3} \frac{Pa^3}{\gamma - 1} = \frac{16\pi \mu r^2 d_r}{3(\gamma - 1)}$$

The constant  $r^2$ d is often referred to as the displacement potential. Figure Al shows the displacement produced by a step-pressure applied to a cavity wall (Ref. 5). The static part of the solution becomes evident at late times. Also shown is the permanent displacement which would be measured by the technique discussed in the report.

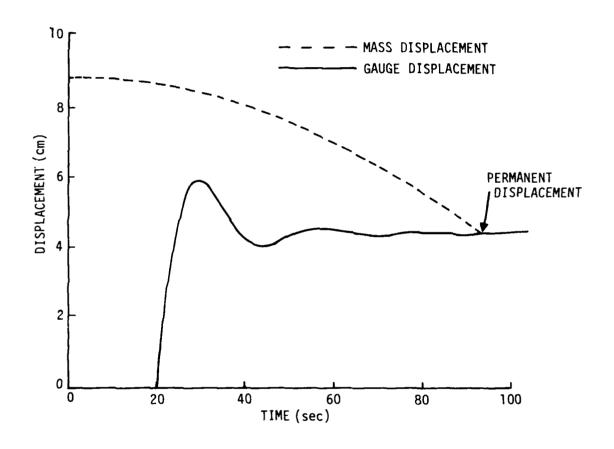


Figure Al. Displacement potential obtained from permanent-displacement measurement.

# APPENDIX B. EFFECTS OF AIR DRAG ON DISPLACEMENT MEASUREMENTS

The relative motion of the drop mass and the air inside the gauge case will apply a drag force opposing the free-fall motion of the weight. In the case of an upward displacement, the flow velocity over the mass will be the ground particle velocity superimposed upon the instantaneous mass velocity. The net force per-unit-area acting on the mass is given by

$$F = 1/2\rho_a (U_p - U)^2 C_d - mg$$
 (B1)

where m = mass per-unit-area of mass,

C<sub>d</sub> = drag coefficient,

U<sub>p</sub> = gauge case velocity,

U = mass velocity,

 $\rho_a$  = air density.

This additional force will affect the time-to-impact and an error in the computed displacement will result. In this Appendix, the error in displacement due to the air drag is estimated.

- a. <u>Assumptions</u>—The equations of motion of the mass can be solved using Eq. (Bl) when  $\mathbf{U}_{p}$  is known; however, the solutions are quite complicated except for special cases. Several simplifying, conservative assumptions will be made and justified by the results. These assumptions are:
  - The particle velocity U<sub>p</sub> is a step function equal to the peak velocity.
  - The drag coefficient  $C_d = 1$ .
  - The relative velocity  $(U_p + U_o)$  is constant over the motion where  $U_o$  is the maximum free-fall velocity in vacuum,  $U_o = \sqrt{2Sg}$ .
  - The mass =  $10 \text{ gm/cm}^2$ .

All of the assumptions are overestimates which will provide an upper bound on the displacement error.

b. Equations of motion--Under these assumptions, the net force F will be constant over the motion, which will be free-fall with an apparent acceleration g' obtained from Eq. Bl:

$$g' = g \left(1 - \frac{1/2\rho_a (U_p - U_o)^2 C_d}{mg}\right)$$
 (B2)

The calculated displacement for the measured time-of-fall  $\mathbf{t}_{m}$  then becomes

$$d_{c} = S - 1/2g't_{m}^{2}$$
, (B3)

and the displacement in the absence of air drag is given by

$$d = S - 1/2qt^2 . (B4)$$

Since

$$d_{C} = U_{p}t_{m}$$
 (B5)

and

$$d = U_{p}t , (B6)$$

the fractional displacement error can be written

$$e = \frac{d - d_{C}}{d}$$

$$= \frac{t - t_{m}}{t}$$
(B7)

where t,  $t_m$  can be obtained by substituting d,  $d_c$  from Eqs. (B5) and (B6) into Eqs. (B3) and (B4) respectively, and solving for t,  $t_m$ :

$$t_{m} = \frac{U_{p}}{g!} \left( \sqrt{1 + 2Sg!/U_{p}^{2}} - 1 \right)$$
 (B8)

$$t = \frac{U_p}{g} \left( \sqrt{1 + 2Sg/U_p^2} - 1 \right)$$
 (B9)

where  $U_0 + U_p$  in Eq. B2 is given by

$$U_o + U_p = U_p \left(1 + \sqrt{2gS/U_p^2}\right)$$

to obtain g'. The parameters in the problem are the gauge size S and the vertical particle or gauge case velocity  $\mathbf{U}_{p}$ , which will be varied in the displacement error computations.

c. Calculations of displacement error-Table Bl shows values of errors obtained by solving Eqs. (B8) and (B9) using the values of S and  $\mathbf{U}_p$  listed in the table. These values probably cover the range for any practical application of the displacement gauge.

The results show that the drag effects are small for the very conservative approximations used in the computations. There are compensating effects which reduce the error. As the particle velocity increases, the drag also increases; however, the time-to-impact decreases, which tends to reduce the drag impulse on the mass.

TABLE B1. PERCENTAGE ERRORS IN DISPLACEMENT DUE TO AIR DRAG

S	U <sub>p</sub> (cr1/sec)			
(cm)	10	100	1,000	10,000
2	0.01	0.01	0.01	0.01
5	0.03	0.04	0.02	0.02
10	0.07	0.08	0.04	0.03
20	0.14	0.16	0.09	0.07
30	0.20	0.24	0.14	0.10

### APPENDIX C. ERROR DUE TO ROTATIONAL MOTION

Throughout the report in the discussion on the operation of the displacement gauge, it has been assumed that gauge motion was purely translational without rotation. However, a gauge imbedded in a heterogeneous medium or in a region of tectonic stress could undergo a rotation superimposed upon its linear motion. \* In addition, gravity can also introduce rotational motion of the field of flow. In this appendix, the error produced in vertical displacement measurements due to rotational motion is analyzed.

In Figure Cl, the initial position of the gauge is shown with the base located along AA. Impact of the mass at P would occur for a vertical displacement V in the absence of rotation. A rotational motion is assumed for a constant radius of curvature r through the angle  $\theta$  to the new base position BB. The motion results in a point of impact at P' with an apparent vertical displacement V'. Also, the rest point P<sub>O</sub> is displaced horizontally by H. The error in the vertical displacement with rotation is defined as the fractional difference between the apparent displacement and the true displacement, namely,

$$E = \frac{\Lambda_{\ell} - \Lambda}{\Lambda}$$

From Figure Cl it can be shown that

 $V' = rtan\theta$  $V = rsin\theta$ 

Personal communication between the author and G. Rawson (Consultant, RDA).

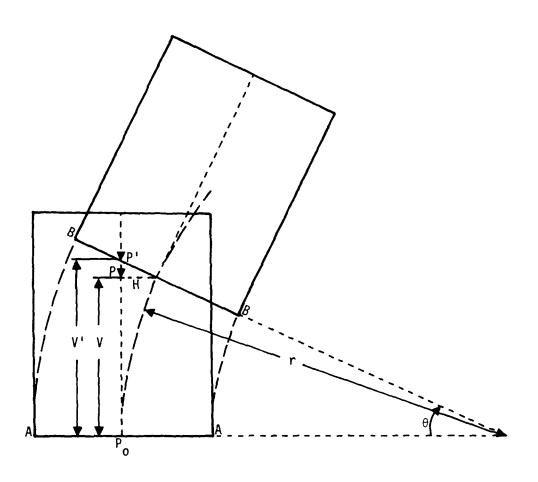


Figure C1. Illustration of effects on ground-motion displacement measurements due to rotational motion.

Hence,

$$\frac{V'-V}{V} = \frac{1-\cos\theta}{\cos\theta}$$

A plot of the percentage difference between the apparent and true vertical displacements is shown in Figure C2. It is interesting to note that the error is independent of r, the radius of curvature. Displacements of interest are expected to be small compared to the radius of curvature in rotational motion, hence the angle of rotation would be correspondingly small. Angular motions to 20 deg produce small errors, as noted in Figure C2.

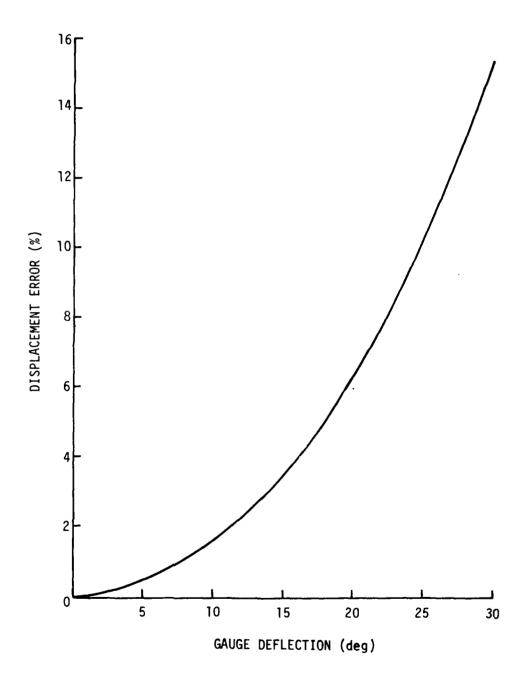


Figure C2. Displacement error due to gauge rotation.

### DISTRIBUTION LIST

#### DEPARTMENT OF DEFENSE

Assistant to the Secretary of Defense Atomic Energy ATTN: Executive Assistant

Defense Intelligence Agency
ATTN: DB-4C, E. O'Farrell
ATTN: DT-1C
ATTN: DT-2

Defense Nuclear Agency 2 cy ATTN: SPSS 4 cy ATTN: TITL

Defense Technical Information Center 12 cy ATTN: DD

Department of Defense Explo Safety Board ATTN: Chairman

Field Command Defense Nuclear Agency ATTN: FCT ATTN: FCPR ATTN: FCTMOF

Field Command Defense Nuclear Agency Livermore Branch ATTN: FCPRL

Field Command Test Directorate Defense Nuclear Agency ATTN: FCTC

NATO School (SHAPE) ATTN: U.S. Documents Officer

Under Secretary of Def for Rsch & Engrg ATTN: Strategic & Space Sys (OS)

# DEPARTMENT OF THE ARMY

Chief of Engineers Department of the Army
ATTN: DAEN-MCE-D
ATTN: DAEN-RDL

Harry Diamond Laboratories Department of the Army

ATTN: CSSA-ADL

ATTN: DELHD-N-P ATTN: 00100 Commander/Tech Dir/TS0

U.S. Army Ballistic Research Labs ATTN: DRDAR-BLT, J. Keefer ATTN: DRDAR-TSB-S ATTN: DRDAR-BLV

ATTN: DRDAR-BLT, W. Taylor U.S. Army Concepts Analysis Agency

U.S. Army Material & Mechanics Rsch Ctr ATTN: Technical Library

#### DEPARTMENT OF THE ARMY (Continued)

U.S. Army Engr Waterways Exper Station ATTN: WESSD, J. Jackson ATTN: J. Strange ATTN: Library ATTN: WESSA, W. Flathau

U.S. Army Materiel Dev & Readiness Cmd ATTN: DRXAM-TL

U.S. Army Missile Command ATTN: RSIC

U.S. Army Mobility Equip R&D Cmd ATTN: DRDME-WC

U.S. Army Nuclear & Chemical Agency ATTN: Library

XVIII Airborne Corps Department of the Army ATTN: F. Ford

#### DEPARTMENT OF THE NAVY

David Taylor Naval Ship R&D Ctr ATTN: Code 1844 ATTN: Code 17 ATTN: Code L42-3

2 cy ATTN: Code 1740.5, B. Whang

Naval Civil Engineering Laboratory ATTN: Code LO8A
ATTN: Code L51, J. Crawford
ATTN: L51, R. Murtha

Naval Electronic Systems Command ATTN: PME 117-21

Naval Facilities Engineering Command ATTN: Code 04B

Naval Material Command ATTN: MAT 08T-22

Naval Research Laboratory ATTN: Code 8440, G. O'Hara ATTN: Code 2627

Naval Sea Systems Command ATTN: SEA-0351 ATTN: SEA-09G53 ATTN: SEA-0322

Naval Surface Weapons Center ATTN: Code R14, I. Blatstein ATTN: Code F31 ATTN: Code R14

Naval Surface Weapons Center ATTN: Tech Library & Info Svcs Br

Office of Naval Research ATTN: Code 474, N. Perrone

#### DEPARTMENT OF THE NAVY (Continued)

Office of the Chief of Naval Operations

ATTN: OP 03EG ATTN: OP 981

Strategic Systems Project Office Department of the Navy

ATTN: NSP-43 ATTN: NSP-272

## DEPARTMENT OF THE AIR FORCE

Air Force Geophysics Laboratory ATTN: LWW, K. Thompson

Air Force Institute of Technology ATTN: Library

Air Force Systems Command ATTN: DLW

Air Force Weapons Laboratory

Air Force Systems Command
ATTN: NTE, M. Plamondon
ATTN: NTES-C, R. Henny
ATTN: SUL

ATTN: DEX

Air University Library Department of the Air Force ATTN: AUL-LSE

Assistant Chief of Staff Intelligence Department of the Air Force ATTN: INT

Deputy Chief of Staff Research, Development, & Acq Department of the Air Force ATTN: AFRDQI

Strategic Air Command Department of the Air Force ATTN: NRI, STINFO Library

## DEPARTMENT OF ENERGY

Department of Energy Albuquerque Operations Office ATTN: CTID

Department of Energy ATTN: OMA/RD&T

Department of Energy Nevada Operations Office ATTN: Mail & Rec for Tech Lib

## OTHER GOVERNMENT AGENCIES

Central Intelligence Agency ATTN: OSWR/NED

Federal Emergency Management Agency ATTN: Assistant Associated Dir

#### DEPARTMENT OF ENERGY CONTRACTORS

Lawrence Livermore National Lab ATTN: Technical Info Dept Library

Los Alamos National Scientific Lab ATTN: MS 364 ATTN: MS 670, J. Hopkins

Sandia National Laboratories Livermore Laboratory
ATTN: Lib & Sec Class Div

Sandia National Lab ATTN: 3141

#### DEPARTMENT OF DEFENSE CONTRACTORS

Aerospace Corp ATTN: Technical Information Services

Applied Research Associates, Inc ATTN: N. Higgins ATTN: J. Bratton

BDM Corp

ATTN: T. Neighbors ATTN: Corporate Library

Boeing Co ATTN: Aerospace Library

California Research & Technology, Inc ATTN: M. Rosenblatt

California Research & Technology, Inc ATTN: D. Orphal

EG&G Wash Analytical Svcs Ctr, Inc ATTN: Library

Electromech Sys of New Mexico, Inc ATTN: L. Piper

Electromech Sys of New Mexico, Inc ATTN: R. Shunk

Eric H. Wang Civil Engineering Rsch Fac ATTN: N. Baum

Geocenters, Inc ATTN: E. Marram

H-Tech Labs, Inc ATTN: B. Hartenbaum

IIT Research Institute
ATTN: Documents Library

Institute for Defense Analyses ATTN: Classified Library

**JAYCOR** 

ATTN: H. Linnerud

# DEPARTMENT OF DEFENSE CONTRACTORS (Continued)

Kaman Sciences Corp ATTN: Library

Kaman Tempo

ATTN: DASIAC

Lockheed Missiles & Space Co, Inc ATTN: T. Geers
ATTN: Technical Library

Lockheed Missiles & Space Co, Inc ATTN: TIC, Library

Lovelace Biomedical & Environmental Rsch Inst, Inc ATTN: D. Richmond

Martin Marietta Corp ATTN: G. Freyer

University of New Mexico ATTN: CERF, N. Baum ATTN: CERF, G. Leigh

Pacific-Sierra Research Corp ATTN: H. Brode

Pacifica Technology ATTN: G. Kent

Physics Applications, Inc ATTN: F. Ford

Physics International Co ATTN: Technical Library ATTN: F. Sauer ATTN: E. Moore

R & D Associates

ATTN: R. Port ATTN: Technical Information Center

ATTN: J. Lewis
ATTN: P. Haas
4 cy ATTN: J. Whitener

#### DEPARTMENT OF DEFENSE CONTRACTORS (Continued)

Science Applications, Inc ATTN: J. Dishon

Science Applications, Inc ATTN: Technical Library

Science Applications, Inc ATTN: J. Cockayne ATTN: W. Layson ATTN: M. Knasel

Southwest Research Institute ATTN: W. Baker ATTN: A. Wenzel

SRI International ATTN: G. Abrahamson ATTN: B. Gasten

Systems, Science & Software, Inc ATTN: Library ATTN: D. Grine

TRW Defense & Space Sys Group ATTN: D. Baer ATTN: Technical Information Center 2 cy ATTN: N. Lipner

TRW Defense & Space Sys Group ATTN: P. Dai ATTN: E. Wong

Weidlinger Assoc, Consulting Engineers ATTN: M. Baron

Weidlinger Assoc, Consulting Engineers ATTN: J. Isenberg

Weidlinger Assoc, Consulting Engineers ATTN: A. Misovec

Plan K

